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1 A mesh-free approach for multiscale modeling in continuum–granular systems

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3

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11

12 ABSTRACT

13 Geotechnical systems often examine interactions that occur between continuum bodies  
14 and granular soils. The systems and interactions can be accurately simulated by using  
15 multiscale coupling approaches. The model for the continuum bodies is often  
16 constructed into a mesh. The meshing however is time consuming for a huge spatial  
17 extent system and if distorted is subject to adjustments. A mesh-free approach can be  
18 used to eliminate these drawbacks. In this study, a mesh-free approach for simulating  
19 continuum–granular systems is presented. This approach combines element-free  
20 Galerkin (EFG) and discrete element (DE) methods to approximate the interactions.  
21 The capabilities of the coupled EFG–DE method are validated through its solving two  
22 example problems: the cantilever beam–disc system and Cundall’s *Nine Disc Test*. The  
23 proposed approach appears to be an efficient and promising tool to model multiscale,  
24 multibody contacting problems.

25 Keywords: meshless, element-free Galerkin, discrete element, coupling, multiscale

26

27

## 28 1. INTRODUCTION

29 Multiscale modeling offers solutions, usually better than the solo scale (i.e., macro- or  
30 microscale) modeling, for geotechnical systems where a huge spatial continuum body  
31 presents in an assembly of granular soils [1, 2]. In these systems, multiscale modeling  
32 enables material response examinations at different scales of resolution, i.e. the  
33 macroscale approximation for the continuum bodies and microscale insight into the  
34 granular media [3]. To combine the multiscale examinations, coupling methods are  
35 applied. A usual coupling method is to develop the finite–discrete element models  
36 (FDEM) [4-9]. For the finite element analyses, meshing, re-meshing, or mesh distortion  
37 sometimes adds to the computational expenses or comprises the simulation accuracy  
38 [10-13]. Similar meshing problems occur to other mesh-based numerical methods, such  
39 as the finite difference [14, 15]. As suggested by Liu and Gu [16], meshless methods  
40 offer choices of solution for these meshing problems.

41 Meshless methods were developed in the 1970s with the intention of reducing  
42 engineers' dependence on meshes. Meshless methods encompass a wide range of  
43 varieties, such as the natural element method [17], the scaled boundary method [18],  
44 and the element-free Galerkin (EFG) method [19]. Not until the 1990s, the EFG method  
45 was developed by Belytschko *et al.* [19] as a tool to predict fracture and crack growth.  
46 In this context, the crack propagation problem is modelled as an extending line. As the  
47 crack develops, the nodal points adjacent to the crack paths will lose their domain  
48 influence to the neighbouring nodes as they are separated. The EFG method defines a

49 grid of nodes which are distributed over the problem domain (i.e, the continuum body),  
50 thus enabling a meshless representation of the body. Based on the nodes distribution, a  
51 shape function (for interpolations) is constructed. Where interpolations (for the inter-  
52 nodes) are needed, the EFG method uses the moving least square (MLS) approximation  
53 to establish algebraic expressions for the shape function. Overall the EFG method  
54 adopts an open, global form [16], that enables its uses in a range of engineering  
55 modeling problems [20-26].

56 The EFG method has been combined with other numerical tools in earlier  
57 studies. Most often, such as [27-29], the EFG method is combined with the finite  
58 element method in order to optimize domain meshing, improve computation efficiency  
59 and increase results accuracy. For example, Ullah *et al.* [30] used the finite element  
60 method to mesh the continuum domain at the outset and converted part of the domain  
61 into EFG nodes where necessary. In some other studies, such as [31-33], the EFG  
62 method can also be combined with the boundary method in order to improve the  
63 solution efficiency and to take the full advantages of the individual methods. These  
64 coupled approaches have proven successful in outperforming corresponding solo  
65 methods, however, are generally applicable to problems of continuum domain. The  
66 approaches are restricted, if not prohibited, from granular systems such as geomaterials,  
67 where the particulate nature is of interest and should be replicated.

68 In this study, the EFG method was coupled with the discrete element (DE)  
69 method to examine interactions that occur between continuum and graular bodies. The

70 EFG and DE modeling was applied to the two bodies respectively. The granular body  
71 comprised an assembly of discrete particles. The continuum–granular interfacial  
72 contacts were detected in terms of algorithms and used to transfer stresses and  
73 deformations between the bodies. The coupling method was validated through it  
74 solving two interesting example problems. To implement the method to the examples,  
75 the coupling was programmed and executed using the MATLAB software package. The  
76 single-platform programming avoids multiple-platform communications which  
77 multiscale modeling often use and prompts the computation efficiency.

78

## 79 2. GOVERNING EQUATIONS

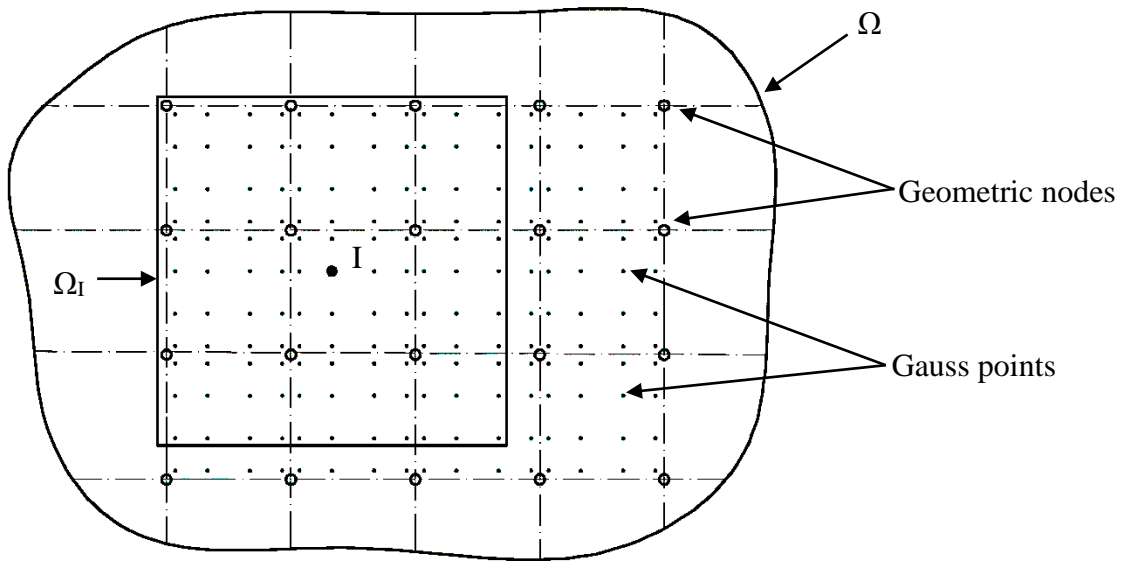
80 In this section, the formulations that are established for the EFG and DE domains in a  
81 two-dimensional (2D) space are presented. The information obtained from the two  
82 domains is communicated at the domains interface, thus updating the resultant contact  
83 forces at each time step. The details of the algorithms developed for the transfer of  
84 forces are discussed. The responses at each of the nodes and particles in respective  
85 domains are examined.

86

### 87 2.1. *Continuum domain*

88 The continuum domain and its boundaries are represented by a grid of nodes. Although  
89 the method of nodes development and the choice of the domain shape are arbitrary, a  
90 grid of nodes in square, as shown in Figure 1, is usually used for simplification. The

91 shape functions can be formed in a local support domain within the problem domain.  
 92 The problem domain and the local support domain are represented by  $\Omega$  and  $\Omega_I$   
 93 respectively. Based on a Gauss quadrature rule [34], Gauss points are distributed in the  
 94 background cell as illustrated so that the locations of the influenced nodes in the local  
 95 domain are identified.



96  
 97 Figure 1. Schematic of the EFG domain presented by nodes and Gauss points.

98  
 99 The MLS approximation, as suggested in [16], is used to construct the shape  
 100 functions. The MLS approximation applies to the local support domain  $\Omega_I$ . As a set of  
 101 equations are to be solved at the point of interest, an ill-conditioned system, where the  
 102 solutions exist but difficult to find, may occur [22, 26]. To address this issue, an  
 103 orthogonal basis function together with the MLS application is used to approximate the  
 104 nodal displacement. Define the nodal displacement trial function,  $u^h(x)$ , as the  
 105 approximation of the actual displacement,  $u(x)$ , at the point of interest. The trial

106 function is written as:

$$u^h(\mathbf{x}) = \sum_{j=1}^m q_j(\mathbf{x}, \bar{\mathbf{x}}) a_j(\bar{\mathbf{x}}) \equiv \mathbf{q}^T(\mathbf{x}, \bar{\mathbf{x}}) \mathbf{a}(\bar{\mathbf{x}}) \quad (1)$$

107 where  $q_j(\mathbf{x}, \bar{\mathbf{x}})$  are the orthogonal basis functions corresponding to the monomial  
 108 basis function  $p(\mathbf{x})$ ,  $a_j(\bar{\mathbf{x}})$  are the coefficients, and  $m$  is the number of elements in the  
 109 monomial basis function. To simplify the coupling framework with the DEM, a linear  
 110 basis function in the 2D domain is created as:

$$\mathbf{p}^T(\mathbf{x}) = [1, x, y] \quad (2)$$

111 By using the Schmidt method [35], the orthogonal basis function is obtained as:

$$q_k(\mathbf{x}, \bar{\mathbf{x}}) = p_k(\mathbf{x}) - \sum_j^{k-1} \alpha_{kj}(\bar{\mathbf{x}}) q_j(\mathbf{x}, \bar{\mathbf{x}}) \quad (3)$$

112 where,  $k=1$  to 3, and the coefficient  $\alpha_{kj}(\bar{\mathbf{x}})$  is expressed as:

$$\alpha_{kj}(\bar{\mathbf{x}}) = \frac{\sum_I^n w_I(\bar{\mathbf{x}}) p_k(x_I) q_j(x_I, \bar{\mathbf{x}})}{\sum_I^n w_I(\bar{\mathbf{x}}) q_j^2(x_I, \bar{\mathbf{x}})} \quad (4)$$

113 where the index  $n$  refers to the nodes number in domain  $\Omega_I$ , and  $w_I(\bar{\mathbf{x}})$  is the weight  
 114 function and usually determined based on the exponential weight function or the conical  
 115 weight function [19]. In this study, the cubic spine weight function [16] is adopted:

$$w(\mathbf{x} - x_I) = w(r) = \begin{cases} 2/3 - 4r^2 + 4r^3 & r \leq 0.5 \\ 4/3 - 4r + 4r^2 - (4/3)r^4 & 0.5 < r \leq 1 \\ 0 & r > 1 \end{cases} \quad (5)$$

116 where  $r = d_I / d_{mI}$ ,  $d_I = |\mathbf{x} - x_I|$ ,  $d_{mI} = d_{max} \times c_I$ ,  $d_{max}$  is the scaling factor, and  $c_I$  can be  
 117 defined as characteristic length of the integration zone that contains the point  $x_I$ . In a  
 118 2D space, the weight function is expressed as



$$w(x - x_I) = w(r_x)w(r_y) = w_x w_y \quad (6)$$

119 where  $r_x$  and  $r_y$  are calculated respectively as

$$r_x = \frac{|x - x_I|}{d_{mx}} = \frac{|x - x_I|}{d_{\max} c_{xI}} \quad (7)$$

$$r_y = \frac{|y - y_I|}{d_{my}} = \frac{|x - x_I|}{d_{\max} c_{xI}} \quad (8)$$

120 where  $d_{mx}$  and  $d_{my}$  are sizes of the support domain  $\Omega_I$ , and  $c_{xI}$  and  $c_{yI}$  are coefficients  
 121 calculated at node  $I$  by searching for nodes to satisfy the base function in both directions.

122 In the Hilbert space span  $q$ , for the selected point  $x$  and weight function  $w$ , the  
 123 orthogonal function  $q_j(x, \bar{x})$  should satisfy the expression as follows:

$$\sum_I^m w(x_I, \bar{x}) q_k(x_I, \bar{x}) q_j(x_I, \bar{x}) = 0 \quad (9)$$

124 where  $m = 3$ ,  $k \neq j$ , and  $k, j = I, \dots, m$ . In terms of the MLS approximation, the difference  
 125 between the trial displacement  $u^h(x)$  and actual displacement  $u(x)$  should be  
 126 minimized. Define the least square function as:

$$\begin{aligned} J &= \sum_{I=1}^n w(x - x_I) [u^h(x, x_I) - u(x_I)]^2 \\ &= \sum_{I=1}^n w(x - x_I) [q^T(x_I, \bar{x}) a_i(\bar{x}) - u(x_I)]^2, \end{aligned} \quad (10)$$

127 Minimizing  $J$ , the coefficients  $a_j(\bar{x})$  are obtained as:

$$a_j(\bar{x}) = \frac{\sum_{I=1}^n w_I(\bar{x}) q_j(x_I, \bar{x}) u_I}{\sum_{I=1}^n w_I(\bar{x}) q_j^2(x_I, \bar{x})}, \quad j = 1 \text{ to } m \quad (11)$$

128 Applying the MLS approximation, we have:

$$u^h(x) = \sum_I^n \phi_I(x) u_I \quad (12)$$

129 Therefore the shape function  $\phi_I(x)$  is defined as:

$$\phi_I(x) = w_I(x) \frac{\sum_j^m q_j(x, x) q_j(x_I, x)}{\sum_I^n w_I(\bar{x}) q_j^2(x_I, \bar{x})} \quad (13)$$

130 The partial derivative of the shape function is expressed as:

$$\phi_{I,k}(x) = w_{I,k}(x) \frac{\sum_j^m q_j(x, x) q_j(x_I, x)}{\sum_I^n w_I(\bar{x}) q_j^2(x_I, \bar{x})} + w_I(x) \frac{A1 - A2}{\left[ \sum_I^n w_I(\bar{x}) q_j^2(x_I, \bar{x}) \right]^2} \quad (14)$$

131 where the subscript 'k' denotes partial derivative to x or y, and parameters A1 and A2

132 are expressed respectively as

$$A1 = \sum_I^n w_I(x) q_j^2(x_I, x) [q_{j,k}(x, x) q_j(x_I, x) + q_j(x, x) q_j(x_I, x)] \quad (15)$$

$$A2 = q_j(x, x) q_j(x_I, x) \left[ \sum_I^n w_{I,k}(x) q_j^2(x_I, x) + 2 \sum_I^n w_I(x) q_j(x_I, x) q_{j,k}(x_I, x) \right] \quad (16)$$

133 Note that  $w_I(x)$ ,  $q_j(x, x)$  and  $q_j(x_I, x)$  are derivable with respect to x.

134

## 135 2.2. Dynamic equation

136 According to Liu and Gu [16], the dynamic equation for node I in the local domain is

137 expressed as:

$$\int_{\Omega_I} \widehat{W}_I (\sigma_{ij,j} + b_i - \rho \ddot{u}_i - c \dot{u}_i) d\Omega = 0 \quad (17)$$

138 where  $\widehat{W}_I$  is the weight function. In a discretized system, the dynamic equation for

139 node I is written as:

$$\mathbf{M}_I \ddot{\mathbf{u}}(\mathbf{t}) + \mathbf{C}_I \dot{\mathbf{u}}(\mathbf{t}) + \mathbf{K}_I \mathbf{u}(\mathbf{t}) = \mathbf{F}_I(\mathbf{t}) \quad (18)$$

140 where  $\mathbf{M}_I$ ,  $\mathbf{K}_I$  are the local mass and stiffness matrix respectively for node I,  $\mathbf{C}_I$  is the

141 corresponding damping matrix,  $\mathbf{F}_I$  is the force acting on node I at time t, and  $\mathbf{u}_I(\mathbf{t})$ ,

142  $\dot{\mathbf{u}}_I(\mathbf{t})$ ,  $\ddot{\mathbf{u}}_I(\mathbf{t})$  are nodal displacement, velocity and acceleration at time t.

143 On the traction boundary  $\Gamma_t$ , the boundary conditions are written as:

$$\boldsymbol{\sigma} \cdot \mathbf{n} = \bar{\mathbf{t}} \quad (19)$$

144 where  $\boldsymbol{\sigma}$  is the stress tensor,  $\mathbf{n}$  is the unit normal to the domain  $\Omega$ , and  $\bar{\mathbf{t}}$  are the  
 145 prescribed tractions. On the displacement boundary  $\Gamma_u$ , the boundary conditions  
 146 become

$$\mathbf{u} = \bar{\mathbf{u}} \quad (20)$$

147 where  $\bar{\mathbf{u}}$  are the prescribed displacements. In order to satisfy the boundary conditions,  
 148 as suggested in Liu and Gu [16], the penalty method (i.e., optimization algorithms) is  
 149 adopted for simplicity and also maintain the symmetrical matrix. By introducing the  
 150 penalty coefficient  $\alpha$  [26], the Galerkin form [16] for a dynamic problem is written as:

$$\int_{\Omega} \delta \mathbf{u}^T \rho \ddot{\mathbf{u}} \, d\Omega + \int_{\Omega} \delta \mathbf{u}^T \mathbf{c} \dot{\mathbf{u}} \, d\Omega + \int_{\Omega} \delta \boldsymbol{\varepsilon}^T \boldsymbol{\sigma} \, d\Omega - \int_{\Omega} \delta \mathbf{u}^T \mathbf{b} \, d\Omega - \int_{\Gamma_t} \delta \mathbf{u}^T \bar{\mathbf{t}} \, d\Gamma + \int_{\Gamma_u} \delta \mathbf{u}^T \alpha (\mathbf{u} - \bar{\mathbf{u}}) \, d\Gamma = 0 \quad (21)$$

151 where  $\delta$  is the test function, and  $\boldsymbol{\alpha} = \begin{bmatrix} \alpha_1 & 0 \\ 0 & \alpha_2 \end{bmatrix}$ . The penalty factors  $\alpha_i$  are usually  
 152 assigned a constant, large, positive number, and this study adopts  $\alpha_i = 10^5 \times E$ . Using Eq.  
 153 (21), the discretized function for a dynamic problem can be developed. The detailed  
 154 process was discussed in Zhang *et al.* [26], and is expressed as:

$$\mathbf{M}\ddot{\mathbf{U}} + \mathbf{C}\dot{\mathbf{U}} + (\mathbf{K} + \mathbf{K}^a)\mathbf{U} = \mathbf{F} + \mathbf{F}^a \quad (22)$$

155 In Eq. (22),  $\mathbf{U}$ ,  $\dot{\mathbf{U}}$  and  $\ddot{\mathbf{U}}$  are global vectors for displacement, velocity and  
 156 acceleration of all of the nodes, respectively;  $\mathbf{M}$  and  $\mathbf{K}$  are respectively the mass matrix  
 157 and stiffness matrix in the problem domain,  $\mathbf{C}$  is the damping matrix,  $\mathbf{F}$  is the global  
 158 external force vector,  $\mathbf{K}^a$  is the global penalty matrix, and the additional force vector  $\mathbf{F}^a$

159 is derived from the boundary conditions. And, these parameters are expanded as:

$$\mathbf{M}_{IJ} = \int_{\Omega} \Phi_I^T \rho \Phi_J d\Omega \quad (23)$$

$$\mathbf{C}_{IJ} = \int_{\Omega} \Phi_I^T c \Phi_J d\Omega \quad (24)$$

$$\mathbf{K}_{IJ} = \int_{\Omega} \mathbf{B}_I^T \mathbf{D} \mathbf{B}_J d\Omega \quad (25)$$

$$\mathbf{K}_{IJ}^a = \int_{\Omega} \mathbf{B}_I^T \alpha \mathbf{B}_J d\Omega \quad (26)$$

$$\mathbf{F} = \int_{\Omega} \Phi_I^T \mathbf{b} d\Omega + \int_{\Gamma_i} \Phi_I^T \bar{\mathbf{t}} d\Gamma \quad (27)$$

$$\mathbf{F}^a = \int_{\Gamma_u} \Phi_I^T \alpha \bar{\mathbf{u}} d\Gamma \quad (28)$$

160 where  $c$  is the damping coefficient, and the other coefficients are defined as follow:

$$\Phi_I = \begin{bmatrix} \phi_I & 0 \\ 0 & \phi_I \end{bmatrix} \quad (29)$$

$$\mathbf{B} = \begin{bmatrix} \phi_{I,1} & 0 \\ 0 & \phi_{I,2} \\ \phi_{I,1} & \phi_{I,2} \end{bmatrix} \quad (30)$$

161 and, for plan stress problems,

$$\mathbf{D} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & (1-\nu)/2 \end{bmatrix} \quad (31)$$

162 for plan strain problems,

$$\mathbf{D} = \frac{E}{(1+\nu)(1-2\nu)} \begin{bmatrix} 1-\nu & \nu & 0 \\ \nu & 1-\nu & 0 \\ 0 & 0 & (1-2\nu)/2 \end{bmatrix} \quad (32)$$

163

164 2.3. Granular domain

165 In the granular domain, the interaction between the particles, or the particles and wall,

166 is determined based on Newton's second law of motion and the force–displacement law.  
 167 The two laws govern the motion of the entities of interest and update the contact force  
 168 based on the displacement. Similar to the dynamic problem described in Eq. (18), in  
 169 the granular domain, as per Cundall and Strack [36], the particle motion is expressed  
 170 as:

$$m_{p,i}\ddot{\mathbf{u}}_i(\mathbf{t}) + c\dot{\mathbf{u}}_i(\mathbf{t}) = \mathbf{F}_i(\mathbf{t}) \quad (33)$$

$$I_i\ddot{\boldsymbol{\theta}}_i(\mathbf{t}) + c^*\dot{\boldsymbol{\theta}}_i(\mathbf{t}) = \mathbf{M}_i(\mathbf{t}) \quad (34)$$

171 where  $m_{p,i}$  is the mass of disc  $i$ ,  $I_i$  is the moment of inertia of disc  $i$ ,  $\dot{\mathbf{u}}_i(\mathbf{t})$  and  $\dot{\boldsymbol{\theta}}_i(\mathbf{t})$   
 172 are respectively the translational and angular velocities for disc  $i$ ,  $c$  and  $c^*$  are global  
 173 damping coefficients for translational and rotational velocities, respectively, and  $\mathbf{F}_i(\mathbf{t})$   
 174 and  $\mathbf{M}_i(\mathbf{t})$  are resultant force and moment at contact, respectively.

175 The DE model uses a set of mechanical elements (e.g., a spring and dashpot) to  
 176 calculate the contact force occurred between two entities (or particles) of interest. One  
 177 of the widely used models is the linear contact, as presented in Figure 2. A finite overlap  
 178 is allowed between the rigid particles to simulate the particle's deformation. The  
 179 dashpot element is used to reflect viscous behavior at contact. The contact force is  
 180 determined in terms of the deformation of these mechanical elements, or the relative  
 181 displacement between the particles. In the model, the normal and shear forces between  
 182 the entities  $i$  and  $j$ ,  $F_{ij}^n$  and  $F_{ij}^s$ , respectively, are calculated as:

$$F_{ij}^n = k_n\Delta n + \beta k_n\Delta n \quad (35)$$

$$F_{ij}^s = k_s\Delta s + \beta k_s\Delta s \quad (36)$$

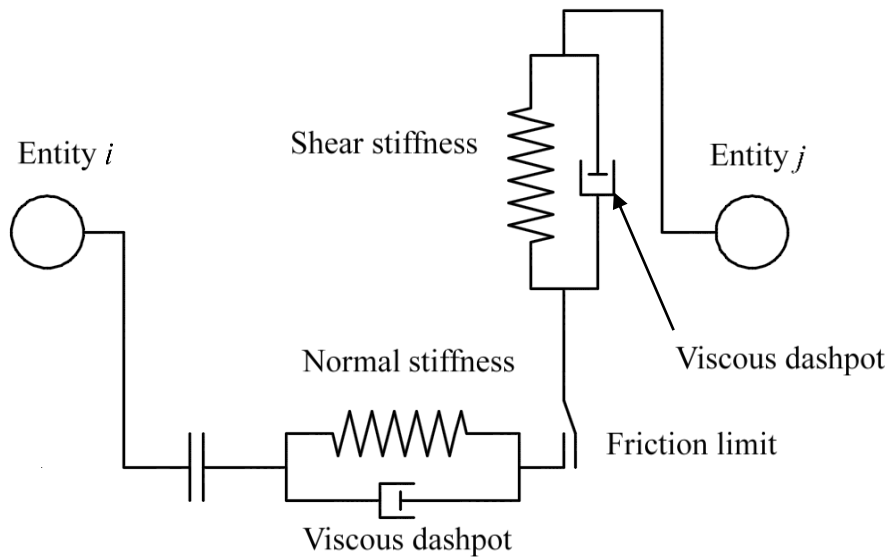
183 where  $k_n$  and  $k_s$  are contact normal and shear stiffness respectively,  $\Delta n$  and  $\Delta s$  are  
 184 relative displacements measured at the normal and shear directions, and  $\beta$  is a damping  
 185 coefficient. To model the stick–slip contact occurred between entities, a Coulomb–  
 186 friction criterion is employed as follows:

$$(F_{ij}^s)_{\max} \leq F_{ij}^n \tan \varphi_u + c \quad (37)$$

187 where  $(F_{ij}^s)_{\max}$  is the maximum value of the shear force,  $\varphi_u$  is the smaller of the  
 188 interparticle friction angles for entities  $i$  and  $j$ , and  $c$  is the smaller of entities' cohesion.  
 189 The moment acting on entity  $i$  is the result of all the shear forces applied at its contacts  
 190 and is expressed as:

$$M_i = \sum_{j=1}^n F_{ij}^s r_i \quad (38)$$

191 where  $r_i$  is the radius of entity  $i$ .



192

193

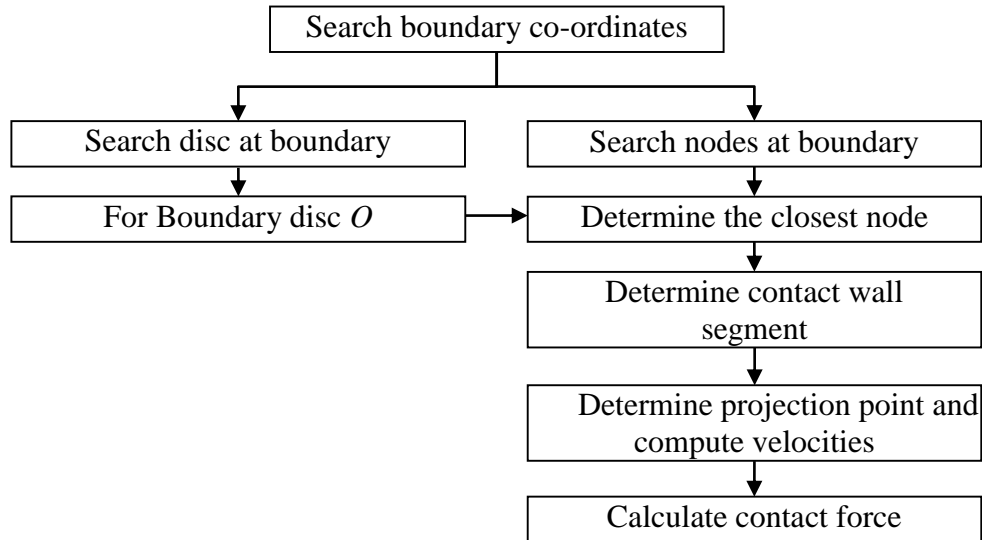
Figure 2. Schematic of the linear contact model used in DEM.

194

195 2.4. *Continuum–granular interface*

196 The continuum–granular domain interface is examined to communicate the force–  
197 displacement relationship between the two domains. The domains interface is modeled  
198 as the disc–wall (or segment) contact, which is commonly used in the FDEM analyses  
199 (e.g. Nakashima and Oida [37]). Specifically, the interface becomes disc–segment  
200 contacts. At each contact, paired disc–segment contact forces are transmitted to the disc  
201 centroid and the nodes of each element at the interface. A bonding strength can be  
202 specified to transmit a tensile strength or a moment. The forces travel to the rest parts  
203 of corresponding domains.

204 A similar concept is used in the EFG–DE domain interface as follows: *a*) detect  
205 the valid contacts between discs (of granular domain) and segments (of continuum  
206 domain), including contact forces and their positions; and *b*) compute the external force  
207 matrix arising from the contacts. The computer flow chart is represented in Figure 3. It  
208 is noted that the EFG–DE method processes the interface force in a way different from  
209 that for the FDEM method. The granular contact force cannot be transmitted directly to  
210 the node forces at the interface, as the shape function obtained does not have Kronecker  
211 delta function property [16]. A new approach is developed to transmit the forces.



212

213 Figure 3. The computer flow chart used to determine the interface force.

214

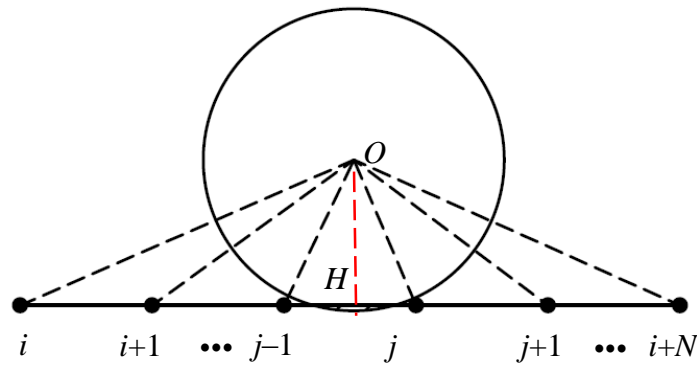
215 *2.4.1. Contact detection*

216 Contacts occur in two forms: the disc–disc contacts in the granular domain and the  
 217 disc–segment on the domains interface. The former type of contact can be detected in  
 218 commercially accessible software packages, e.g. the *PFC*, or an open source code such  
 219 as *Escript* [38] or *Yade*. These packages however are not established to readily detect  
 220 the disc–segment contacts, or otherwise have to use a bridging scheme [39] to  
 221 communicate contact detections across the domains. There are algorithms [11, 12, 40]  
 222 developed to detect finite–discrete element interfacial contacts. These algorithms,  
 223 however, are not applicable to the EFG–DE interface and a separate approach is  
 224 required. To these ends, we developed contact detection algorithms in terms of Muth *et*  
 225 *al.* [41] and programmed the algorithms on the *MATLAB* platform.

226 To detect the EFG–DE interfacial contacts, the first step is to gather location



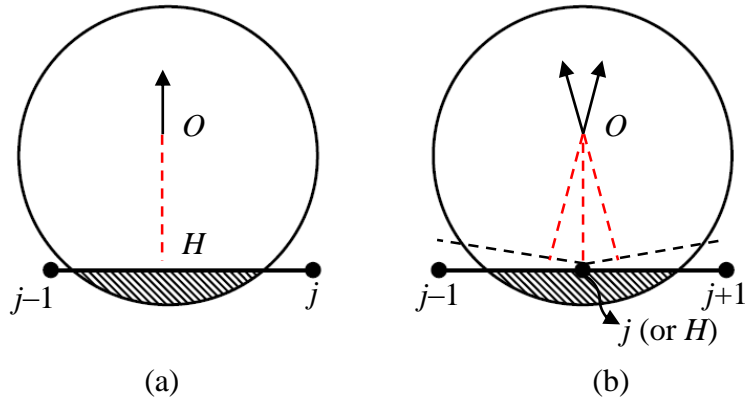
227 information for the nodes and discs adjacent to the interface. Figure 4 illustrates disc  $O$   
 228 and nodes  $i$  to  $i+N$  which contact and sit on the interface. To detect the disc–segment  
 229 contact, the following subroutines are executed: a) Calculate  $d_{i,o}$ , the distance between  
 230 centroid  $O$  and node  $i$ , where  $i=i, \dots, i+N$ ; b) Determine the minimum distance  $(d_{i,o})_{min}$   
 231 and the corresponding node number  $j$ ; c) Calculate distances  $d_{j-1,o}$  and  $d_{j+1,o}$ ; d)  
 232 Determine the interface segment. The segment is section  $(j-1, j)$  if  $d_{j-1,o} > d_{j+1,o}$ , or  
 233 section  $(j, j+1)$  if  $d_{j-1,o} < d_{j+1,o}$ . If  $d_{j-1,o} = d_{j+1,o}$ , the segment is dependent on the distance  
 234 between the centroid and the segments of interest which is discussed in the next  
 235 paragraph; e) Calculate  $d_{H,o}$ , the distance between centroid  $O$  and point  $H$ . Line  $OH$  is  
 236 drawn normal to the segment determined in Step d); and f) Calculate the velocity at  
 237 point  $H$  based on the shape function of this segment, and the velocity of nodes  $j$  and  $j-$   
 238 1 based on the EFG method.



241 Figure 4. An illustration of disc position and boundary line segments.

242 The next step is to determine the contact geometric primitives. In the FDEM  
 243 coupling work, Zang *et al.* [11] categorized the contact geometric primitives into  
 244 particle–facet, particle–edge and particle–vertices problems. These contacts are not

245 suitable to the EFG–DE coupling. Instead, two types of disc–segment contacts are  
 246 discussed, as shown in Figure 5(a) and (b) respectively. Figure 5(a) shows the particle–  
 247 segment contact where no nodes sit within the interface segment. Figure 5(b) shows the  
 248 particle–point contact where disc  $O$  contacts node  $j$ . At the particle–point contact, the  
 249 segment  $(j-1, j+1)$  deforms into two sub-segments,  $(j-1, j)$  and  $(j, j+1)$ . In this case, the  
 250 contact force will be doubled. To eliminate this error, distance  $d_{H,O}$  is replaced by  $d_{j,o}$   
 251 in Step  $f$ ) in the contact detection subroutine.



252  
 253  
 254 Figure 5. Schematic of disc–segment contact: (a) particle–segment contact, and (b)  
 255 particle–point contact.

256 Based on the finite difference method, the discrete equation used to calculate the  
 257 increment of disc–segment force is written as:

$$\Delta F_{HO}^n = k_n (v_O - v_H) \Delta t + \beta k_n (v_O - v_H) \Delta t \quad (39)$$

$$\Delta F_{HO}^s = k_s (v_O - v_H) \Delta t + \beta k_s (v_O - v_H) \Delta t \quad (40)$$

258 where  $v_o$  and  $v_H$  are the average velocities of centroid  $O$  and point  $H$  on the segment  
 259 during time step  $\Delta t$ . The velocity of  $v_H$  is expressed as:

$$v_H = v_i + \frac{l_{i,H}}{l_{i,i+1}}(v_{i+1} - v_i) \quad (41)$$

260 or, if  $d_{j-1,O} = d_{j+1,O}$ , is simplified into

$$v_H = v_i \quad (42)$$

261 where  $v_i$  and  $v_{i+1}$  are the velocity of nodes  $i$  and  $i+1$  respectively,  $l_{i,H}$  is the distance

262 between nodes  $i$  and  $H$ , and  $l_{i,i+1}$  is the distance between nodes  $i$  and  $i+1$ .

263

#### 264 2.4.2. Determination of contact force

265 In finite element modeling, contact forces are usually acted as a point load on the

266 boundary nodes [42]. This point-load approach cannot be directly applied to the

267 boundary nodes in the EFG domain, which otherwise invalidates the use of MLS

268 approximation [19]. Alternatively, each load is regarded as a distributed traction, and

269 multiple tractions are superposed. According to Zuohui [43], if a point load  $F$  acts at

270 position  $(x_0, y_0)$  on interface  $\Gamma_i$  as shown in Figure 6, the following equation is obtained:

$$\int_{\Gamma_i} \Phi_i^T t_i d\Gamma = \int_{\Gamma_i} \Phi_i^T F_i \delta(x - x_i) d\Gamma = \Phi_i^T F_i \quad (43)$$

271 where  $\delta$  is the Dirac delta function. Assuming a total of  $N$  point loads act on the

272 boundary, the superposed traction is expressed as:

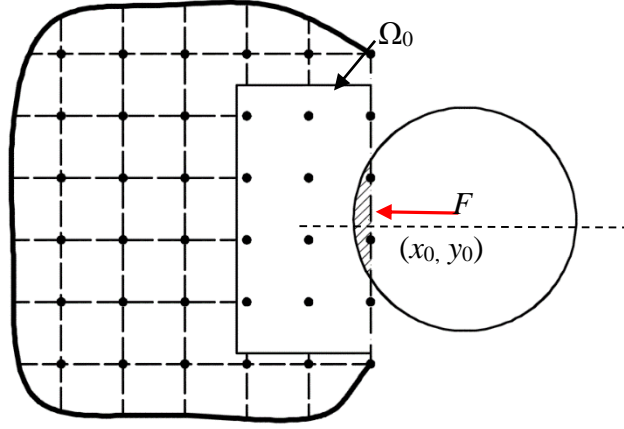
$$\bar{t}(x) = \sum_{i=1}^N t_i = \sum_{i=1}^N F_i \delta(x - x_i) \quad (44)$$

273 Eq. (43) then becomes as

$$\int_{\Gamma_i} \Phi_i^T \bar{t} d\Gamma = \sum_{i=1}^N \Phi_i^T F_i \quad (45)$$

274 Substituting Eq. (45) to Eq. (27) leads to:

$$\mathbf{F} = \int_{\Omega} \Phi_{\mathbf{I}}^T \mathbf{b} d\Omega + \sum_{i=1}^N \int_{\Gamma_i} \Phi_{\mathbf{I}}^T F_i \delta(x - x_i) d\Gamma = \int_{\Omega} \Phi_{\mathbf{I}}^T \mathbf{b} d\Omega + \sum_{i=1}^N \Phi_{\mathbf{I}}^T F_i \quad (46)$$



275

276

Figure 6. The interaction between a disc and an EFG domain.

277

278 In Eq. (46), the external force  $F$  acting on domain  $\Omega$  contains two components:

279 the body force such as the gravity, and the point load. The latter part of the equation

280 refers to the following physical meaning: when a point load  $F_i$  acts at point  $(x_0, y_0)$  on a

281 continuum boundary, this load is distributed to the surrounding points in the local

282 supporting domain  $\Omega_{\mathbf{I}}$  based on shape function  $\phi_i(x)$  which is determined by Eq. (13).

283 The supporting domain area may be affected by the chosen domain scaling factor  $d_{\max}$ .

284

### 285 3. TIME INTEGRATION

286 In the EFG–DE simulation, the force–displacement relationship is discretized into finite

287 time steps. To enable a converge of the simulation, the value of the time steps is properly

288 determined to ensure the algorithms are stable in both DE and EFG domains. This

289 section describes the method developed to determine the time step and to present the

290 corresponding governing equations in the two domains.

291

### 292 3.1. Time step

293 A time step is determined either explicitly or implicitly. Belytschko *et al.* [44] discussed  
294 the differences between the explicit and implicit methods and suggested that the choice  
295 of method should be determined in terms of the governing equations, smoothness of  
296 data, and material response to examine. In the discrete element analysis, the central  
297 difference method is often used [36]. This method guarantees numerical stability so that  
298 each time step does not exceed the critical time step in the explicit time scheme. Also  
299 when the particle number increases, the implicit time schemes may require solving  
300 multiple matrices at each time step, which significantly increases the processing time  
301 [45]. Due to the above reasons, one common method in the coupled model is to  
302 determine the time step using explicit–explicit schemes [12, 40], which is expressed as:

$$\Delta t \leq \min(\Delta t_1, \Delta t_2) \quad (47)$$

303 where  $\Delta t_1$  and  $\Delta t_2$  are the minimum time steps in the continuum and granular domains,  
304 respectively. To optimize the time step determination, Elmekati and El Shamy [46]  
305 suggested to use the predictor–corrector method, a two-staged iterative process. This  
306 method arises from the fact that  $\Delta t_2$  is usually much less than  $\Delta t_1$ . Therefore the time  
307 step in the main routine is expressed as  $\Delta t_1$ :

$$\Delta t = \Delta t_2 = n\Delta t_1 = n\sqrt{\frac{m_p}{K}} \quad (48)$$

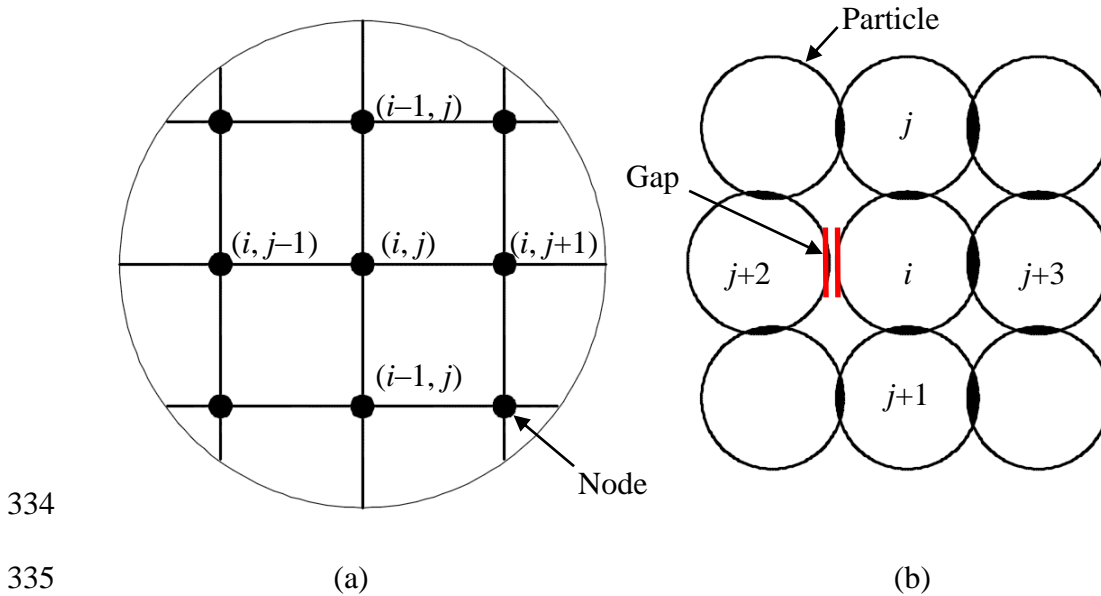
308 where  $n$  is an integer,  $m_p$  is the particle mass, and  $K$  is the contact spring stiffness.

309 In this present study, an explicit–implicit time integration scheme was adopted.  
310 In the continuum domain, the iterations are stable due to the advantages of the implicit  
311 method, if relatively a small- to medium time step increment is used [26]. Therefore,  
312 time steps only need to be determined in the DE domain. Also, the calculation is  
313 consistent in the combined model because the results of the DE simulation can be  
314 transmitted to the EFG domain at each time step. In this context, important information  
315 such as the contact detection on the interface should be attained, so that the conditions  
316 at each node and particle can be examined explicitly while executing major iterations.  
317

### 318 *3.2. Partial difference solution*

319 Difference method is used to discretize the time domain to solve governing equations.  
320 The governing equations relating to accelerations, velocities and displacements arising  
321 from the force acting on the two domains are updated at each time step. It is noteworthy  
322 that the governing equations for the two domains are solved in different processes. The  
323 differences are illustrated in Figure 7. In the EFG domain, the governing equations are  
324 solved based on a matrix, which arises from the nature of continuum body. In the DE  
325 domain, stiffness matrix dimensions may vary in different steps as some particles may  
326 not in contact as shown in Figure 7 (b). Therefore, the contact conditions need to be  
327 determined at the end of each step. It is computationally expensive to compute a  
328 stiffness matrix at each loop. Where appropriate the responses of individual particles  
329 are examined, which avoids excessive iterations of the stiffness matrix. In the EFG

330 domain, the nodes are numbered sequentially and the displacement, velocity and  
 331 acceleration are obtained in matrices (i.e.  $\mathbf{U}$ ,  $\dot{\mathbf{U}}$  and  $\ddot{\mathbf{U}}$  respectively). In the DE  
 332 domain, the displacement, velocity and acceleration are calculated for disc  $i$ , i.e.,  $u_i$ ,  
 333  $\dot{u}_i$  and  $\ddot{u}_i$  respectively.



336 Figure 7. Schematic illustrating EFG-DE domains: (a) EFG domain with difference  
 337 nodes, and (b) DE domain with particles at contact.

338  
 339 Table 1 presents the sets of governing equations used in the EFG and DE  
 340 domains respectively. These equations demonstrate the motions occurred in a time step  
 341 increment from  $t$  to  $t+\Delta t$ . The two sets of equations are tabularised to compare the  
 342 difference in conception when computing nodes (or discs) motion. Specifically, in the  
 343 continuum domain, an external force matrix is a major target; in the granular domain,  
 344 internal disc-disc contact forces are computed to provide the force-displacement

345 relationship. Regarding the equation solving processes, the continuum domain uses the  
346 Taylor expansion to obtain the recurrence relationship at the end of the time increment.  
347 The granular domain uses the central difference method and determines the velocity at  
348  $t + \frac{\Delta t}{2}$  which is known as the average speed during a time step increment. Extra rolling  
349 behavior at disc  $i$ , such as rolling angle  $\theta_i$ , rolling velocity  $\dot{\theta}_i$ , and rolling acceleration  
350  $\ddot{\theta}_i$ , was added in the DE analysis.



Table 1 Governing equations to depict the motion of elements in EFG and DE domains.

	EFG domain	DE domain
Displacement	$\mathbf{U}_{t+\Delta t} = \frac{\mathbf{F}_{t+\Delta t} + \mathbf{M}(\alpha_1 \mathbf{U}_t + \alpha_2 \dot{\mathbf{U}}_t + \alpha_3 \ddot{\mathbf{U}}_t)}{\alpha_1 \mathbf{M} + \bar{\mathbf{K}}} \quad (49)$	$(\mathbf{u}_i)_{t+\Delta t} = (\mathbf{u}_i)_t + (\dot{\mathbf{u}}_i)_{t+\Delta t/2} \times \Delta t \quad (50)$
		$(\theta_i)_{t+\Delta t} = (\theta_i)_t + (\dot{\theta}_i)_{t+\Delta t/2} \times \Delta t \quad (51)$
Velocity	$\dot{\mathbf{U}}_{t+\Delta t} = \dot{\mathbf{U}}_t + (1 - \beta_1) \Delta t \ddot{\mathbf{U}}_t + \beta_1 \Delta t \ddot{\mathbf{U}}_{t+\Delta t} \quad (52)$	$(\dot{\mathbf{u}}_i)_{t+\Delta t/2} = (\dot{\mathbf{u}}_i)_{t-\Delta t/2} + (\ddot{\mathbf{u}}_i)_t \times \Delta t \quad (53)$
		$(\dot{\theta}_i)_{t+\Delta t/2} = (\dot{\theta}_i)_{t-\Delta t/2} + (\ddot{\theta}_i)_t \times \Delta t \quad (54)$
Acceleration	$\ddot{\mathbf{U}}_{t+\Delta t} = \alpha_1 (\mathbf{U}_{t+\Delta t} - \mathbf{U}_t) - \alpha_2 \dot{\mathbf{U}}_t - \alpha_3 \ddot{\mathbf{U}}_t \quad (55)$	$(\ddot{\mathbf{u}}_i)_t = \left( \sum_{j=1}^N F_{ij} / m_{p,i} \right) \Delta t \quad (56)$
		$(\ddot{\theta}_i)_t = \left( \sum_{j=1}^N M_{ij} / I_i \right) \Delta t \quad (57)$
Force	External forces determined in terms of Eqs. (27–28)	$(F_{ij})_{t+\Delta t} = (F_{ij})_t + k(\dot{\mathbf{u}}_i)_{t+\Delta t/2} \Delta t + \beta k(\dot{\mathbf{u}}_i)_{t+\Delta t/2} \Delta t \quad (58)$
		$(M_{ij})_{t+\Delta t} = (M_{ij})_t + k_s \{ (\dot{\mathbf{u}}_i)_{t+\Delta t/2} s \} r_i \Delta t \quad (59)$

353 where in Eq (49), the parameters are respectively expressed as:

$$\bar{\mathbf{K}} = \mathbf{K} + \mathbf{K}^\alpha \quad (60)$$

$$\alpha_1 = \frac{2}{\beta_2 \Delta t^2} \quad (61)$$

$$\alpha_2 = \frac{2}{\beta_2 \Delta t} \quad (62)$$

$$\alpha_3 = \frac{1}{\beta_2} - 1 \quad (63)$$

354 where, two constants  $\beta_1=1.5$  and  $\beta_2=1.6$  are used as the Newmark parameters as per  
355 [26].

356

#### 357 4. EXAMPLE PROBLEMS

358 Two example problems were examined and solved using the developed EFG–DE  
359 method, aiming to validate this coupling method. The first example problem is to assess  
360 a cantilever beam which is subjected to a disc acting at the end of the beam. The second  
361 example problem is developed based on the *Nine Disc Test* [36]. The test reproduces a  
362 bi-axial test on an assembly of nine discs. Contact force evolution and stress distribution  
363 between the discs are estimated. The two example problems consider multi-body  
364 interactions, but involve less number of nodes or discs than required in other large-scale  
365 problems. This means the computational costs are affordable, and these special settings  
366 satisfy the aim of developing and validating the EFG–DE method. The deformation, on  
367 both continuum and granular bodies, however, is executed in a large scale so that the  
368 advantages of the mesh free method can be demonstrated and confirmed.

369

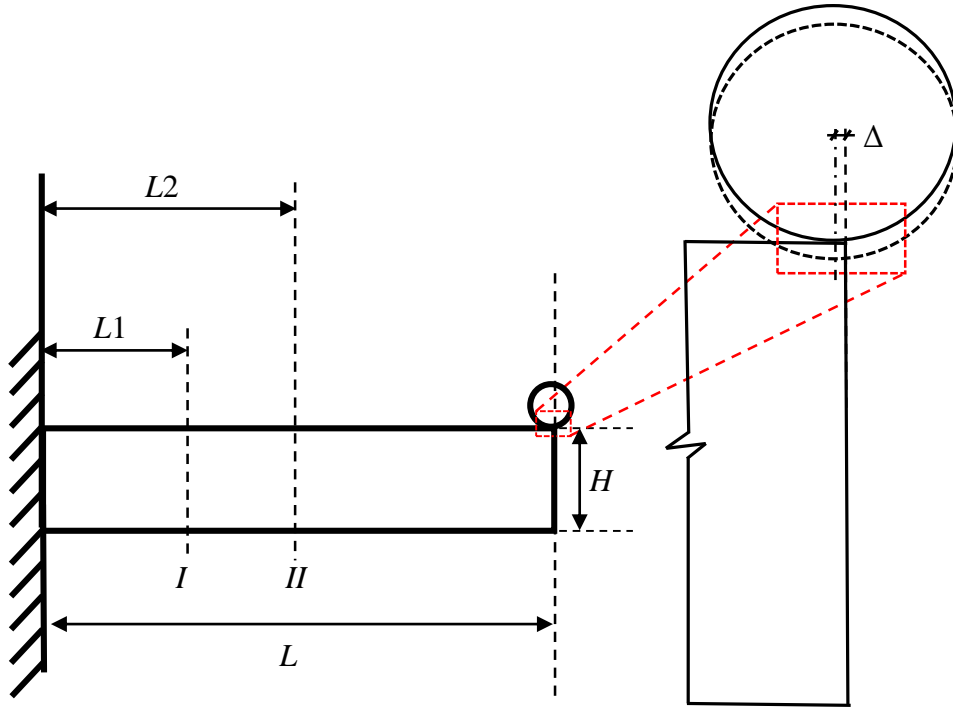
#### 370 *4.1. Example 1*

371 This section presents a study on the dynamic interaction that occurs between a disc and  
372 a cantilever beam. The EFG–DE method is applied to the example problem, and the  
373 numerical results are compared with the analytical solutions developed for the same  
374 example problem.

375

##### 376 *4.1.1. Problem description*

377 In this example problem, the cantilever beam is fixed to a rigid surface, and the disc sits  
378 on the other end, as shown in Figure 8. The beam measures  $1 (L) \times 0.2 (H) \times 0.025 (D)$   
379 m. The material density of the beam is  $\rho_b = 2,000 \text{ kg/m}^3$ . The radius of the disc is  $r =$   
380  $0.05 \text{ m}$ , and its density is  $\rho_d = 1,000 \text{ kg/m}^3$ . It was assumed that the material of the beam  
381 exhibits linear elastic behavior with Young's modulus  $E = 2.1 \times 10^8 \text{ Pa}$  and Poisson's  
382 ratio  $\nu = 0.3$ , and that the disc material is simulated with the linear contact model with  
383 stiffnesses of  $k_n = k_s = 10^6 \text{ N/m}$ . The system is assumed un-damped (i.e., damping  
384 coefficient is zero).



385

386 Figure 8. Schematic of the disc falling down against the end of the cantilever beam.

387

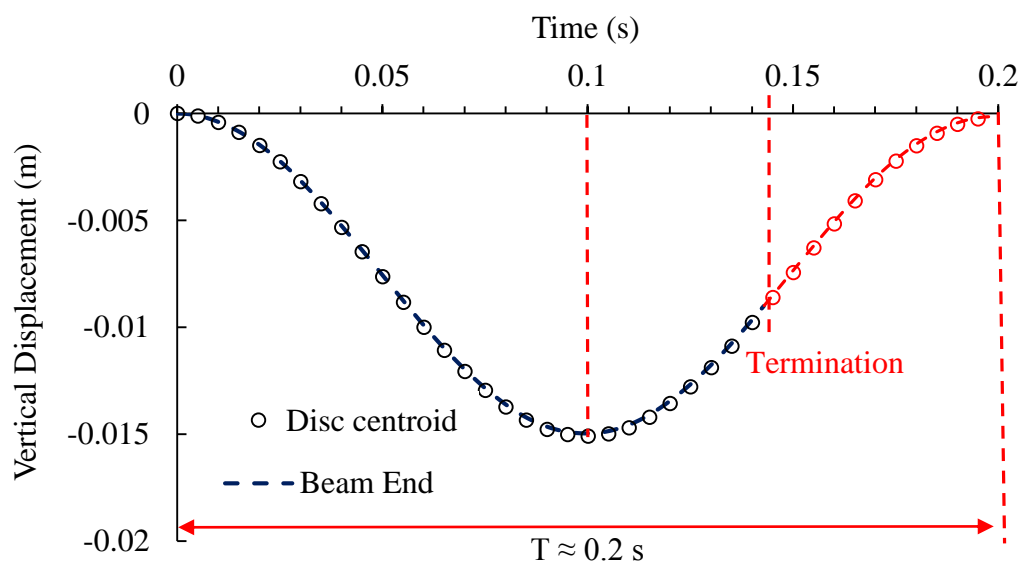
388 In the simulation, the beam is discretized into a node arrangement of  $20 \times 4$ . The  
 389 node grid is refined by a  $4 \times 4$  Gauss quadrature scheme. At time  $t = 0$ , the beam is at  
 390 rest, and its upper-right boundary is in contact with the disc edge (no overlap or  
 391 deformation). The disc centroid sits at a small distance  $\Delta = 10^{-3}$  mm inward from the  
 392 beam end, to ensure that the centroid falls inside the boundary of the beam. When  $t$   
 393 increases, the disc goes down under the gravity and penetrates the boundary of the beam.  
 394 Meanwhile, the beam displaces, in particular at its end, forming a convex profile,  
 395 prompting the disc to fall out.

396

#### 397 4.1.2. Termination condition

398 The termination condition was determined in terms of the trajectory of the disc centroid.

399 The centroid tends to move outward when the beam is bent downward. Where the  
 400 projection of the centroid falls out of the boundary of the beam, the interaction between  
 401 the disc and beam becomes unstable and the simulation terminates. To determine the  
 402 fall-off moment, the vertical displacements of the beam end and the disc centroid are  
 403 captured and plotted with time as shown in Figure 9. Where the two displacement  
 404 values disagree, the corresponding time is when the fall-off occurs. It is shown that the  
 405 corresponding time point is  $t = 0.1431$  s. It is noted that excellent agreement is obtained  
 406 between the two displacement curves before this fall-off time point is reached, thus  
 407 demonstrating the stability of the simulation.



408  
 409 Figure 9. Displacement profile for the end of the beam and the centroid of the disc.

410

#### 411 4.1.3. Model validation

412 The numerical results are compared with the analytical solutions developed for the  
 413 same example problem. The problem was solved in a plane stress condition—a point

414 load  $P$  acting at the upper right corner of the beam. According to Euler–Bernoulli beam  
415 theory, the axial stress,  $\sigma_{11}$ , and the deflection of the beam,  $w_x$ , are respectively  
416 expressed as:

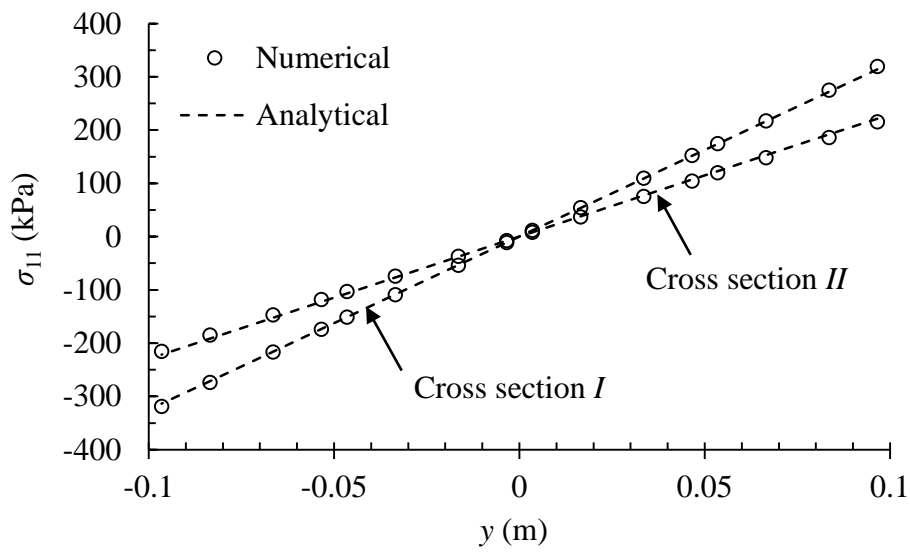
$$\sigma_{11} = \frac{P(L-x)y}{I_m} \quad (64)$$

$$w(x) = \frac{Px^2(3L-x)}{6EI_m} \quad (65)$$

417 where  $(x, y)$  is the coordinate of the cross section of interest, and  $I_m$  is the moment of  
418 inertia of the beam.

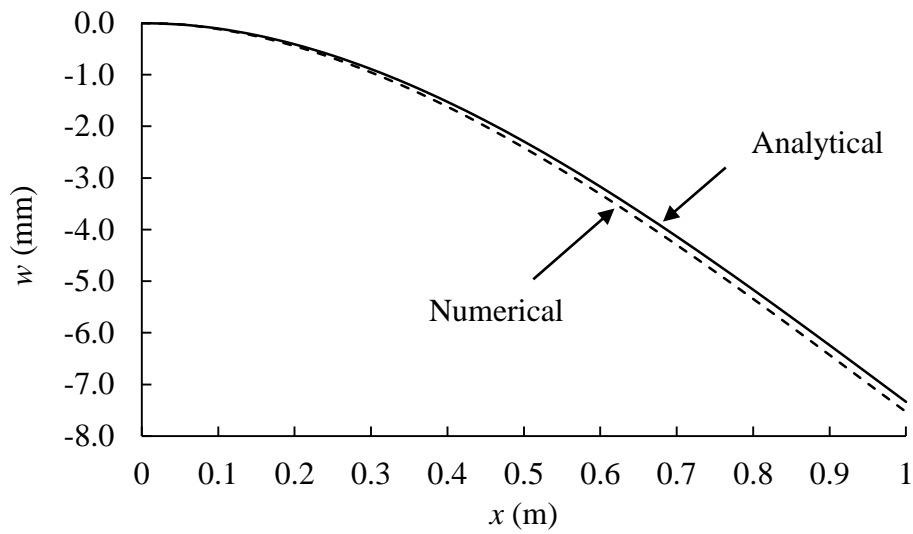
419 The axial stress profiles at two cross sections  $I$  at  $L_1 = 0.3$  m and  $II$  at  $L_2 = 0.5$   
420 m as shown in Figure 8 are obtained. For the simulation results, the axial stress at the  
421 same cross sections is captured. But, due to the beam acting without damping, the  
422 results where the beam is in its minimal acceleration  $t = T/4$ , are used. The results are  
423 presented in Figure 10. The axial stress is plot as a function of the vertical depth  $y$  for  
424 both the simulation and analytical results. At either of the cross sections, excellent  
425 agreement between the simulation and analytical results is obtained. Similarly  
426 satisfactory agreement is obtained for the deflection profile of the beam, as presented  
427 in Figure 11. The results agreement verifies the capability of the EFG–DE model in  
428 simulating the dynamic response of the beam. Furthermore, the orthogonal basis  
429 function was used in the iterations, and this function avoids the occurrence of any ill-  
430 conditioned problems. The similar advantage in simulation stability is obtained due to  
431 the uses of the explicit–implicit algorithm for the time step and the penalty method for

432 the boundary conditions.



433

434 Figure 10. Axial stress profile plot at two cross sections of the cateliber beam.



435

436 Figure 11. Deflection profile of the cantilever beam.

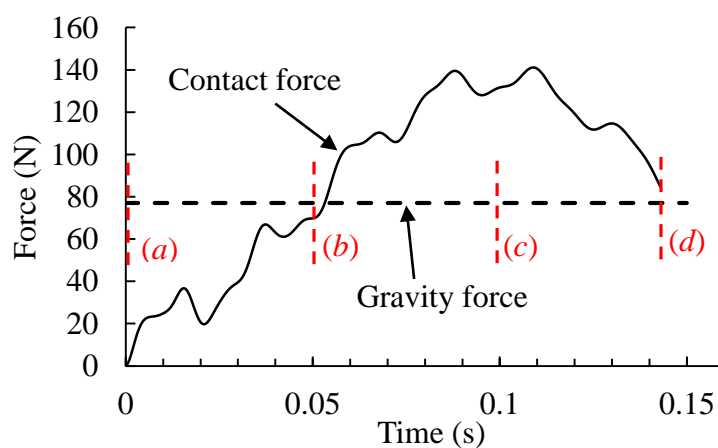
437

#### 438 4.1.4. Variation of contact force with time

439 When the beam is subjected to a dynamic vibration, the contact forces acting on the

440 boundary change over time. The results of the contact forces are provided in Figure 12.

441 In the figure, four critical time steps are identified:  $t = 0.001$ ,  $0.05$ ,  $0.1$ , and  $0.143$  s,  
 442 which correspond to points (a), (b), (c), and (d) respectively. It is shown that the contact  
 443 force gradually increases with time at the early stage of the test. At  $t = 0.05$  s where the  
 444 contact force equals the gravity force, the acceleration becomes zero, and then negative  
 445 when the contact force exceeds the gravity. In the meantime, the disc velocity gradually  
 446 decreases, but the contact force grows at a similar gradient. The contact force attains  
 447 the peak value when  $t = 0.1$  s, and at this moment, the disc attains the maximum  
 448 displacement and penetration into the beam. After the peak point, the penetration  
 449 releases gradually and the contact force goes down. At  $t = 0.143$  s the contact force is  
 450 less than the gravity, and the disc falls off the end of the beam.



451

452 Figure 12. Development of disc-segment contact force over time.

453

#### 454 4.2. Example 2

455 Example 2 was adapted from the *Nine Disc Test* [36]. In the original test, two pairs of  
 456 plates were used to compress an assemblage of 9 discs. The plates were assumed ideally  
 457 rigid. In example 2, the plates were allowed to deform to avoid the rigid body



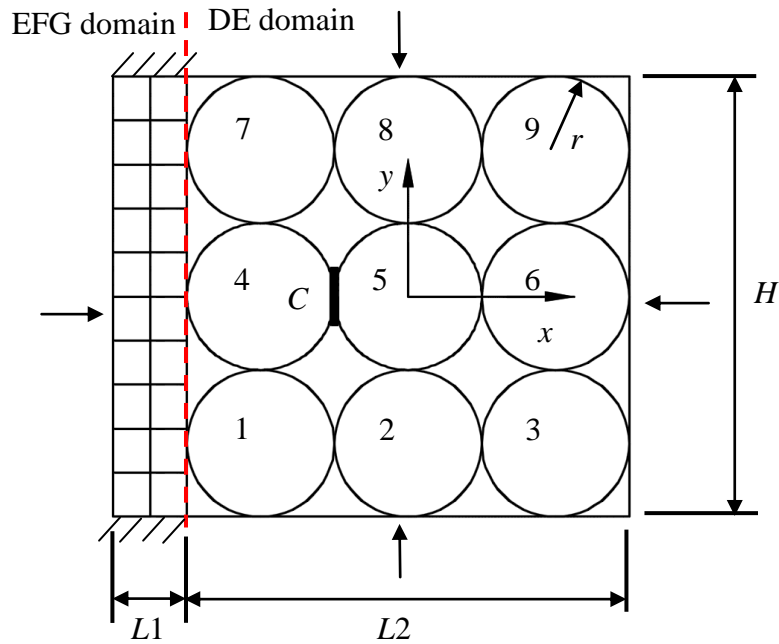
458 assumption. The *Nine Disc Test* is employed because of the following reasons: *a)* this  
459 test is designed to record the single contact force occurred between the discs and walls  
460 (or segments), enabling force gauging; *b)* the test can capture the effects of the plate  
461 deformation on the contact force; and, *c)* the test uses a small number of discs and  
462 facilitates contact detection and simulation in a short time period.

463

#### 464 *4.2.1. Problem description*

465 An assembly of nine discs is sandwiched by two pairs of plates, as shown in Figure 13.  
466 The setup remains the same as in Cundall and Strack [36], except the left-hand side  
467 plate which is replaced with a deformable strip plate. This strip plate dimensions are 50  
468 ( $L1$ )  $\times$  300 ( $H$ )  $\times$  1 ( $D$ ) units, which enables a plane-stress scenario. As per Cundall and  
469 Strack [36], no physical unit but a number is provided to the properties of the setup or  
470 elements. Specifically, the radii are 50 units, the density is 1000 units, and the normal  
471 and shear stiffness are  $k_n = k_s = 1.35 \times 10^9$  units for the linear contact model used for the  
472 discs. In the DE domain, the object wall is not assigned physical properties such as  
473 Young's modulus, Poisson's ratio or density. However, in the EFG domain (i.e., the strip  
474 plate), the material properties are specified in order to constitute a motion. These  
475 properties include Young's Modulus of  $2.1 \times 10^{14}$  units, Poisson's ratio of 0.3, the  
476 density of 2000 units. The plates were assumed to be undamped ( $c=0$  in Eqs. (24)), and  
477 fixed at the top and bottom boundaries. To cross check the capacity of the proposed  
478 coupling method, we also simulated this example by using an FE-DE coupling method.

479 The EFG domain was replaced with the FE domain with a mesh coinciding with the  
 480 background grid of EFG domain. The setting for DE domain remains the same as  
 481 provided in Figure 13.



482

483 Figure 13. The nine disc test performed with deformable boundaries.

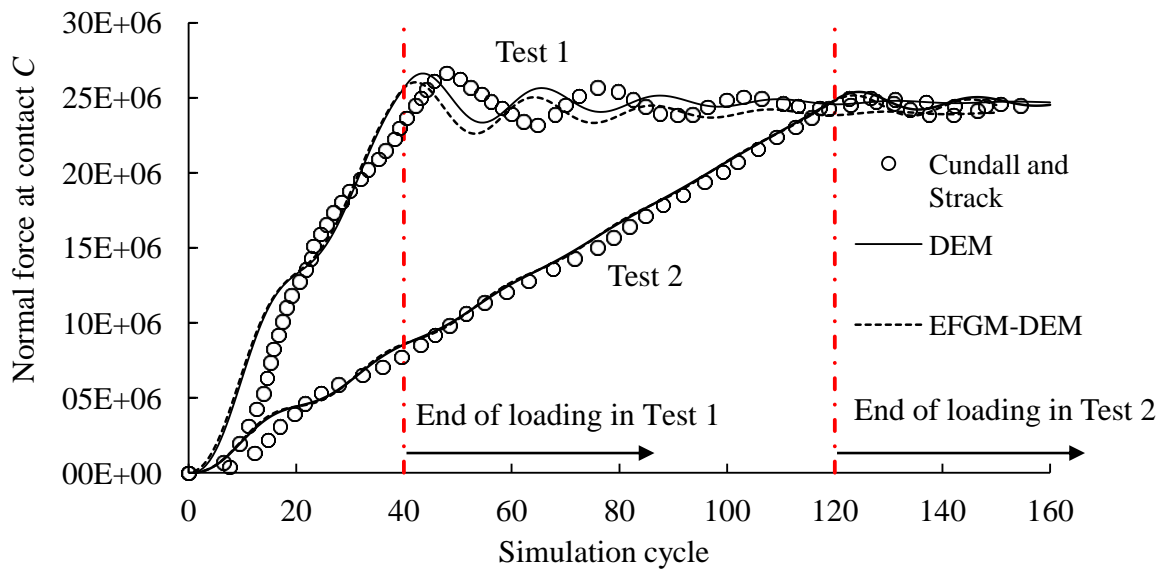
484

485 *4.2.2. Model validation*

486 In the simulation, the assembly of the nine discs is subjected to the bi-axial compression  
 487 provided by the two pairs of plates. Two tests were performed. In Test 1, the plates  
 488 travel at a velocity of 0.12 units and stop after 40 cycles. In Test 2, the velocity reduces  
 489 to 0.04 units, but the plates continue to move until the 120<sup>th</sup> cycle. As per Cundall and  
 490 Strack [36], both simulations continue to the 150<sup>th</sup> cycle and use a time step  $\Delta t =$   
 491 0.01525 units and damping coefficient of 0.1. The continuum domain (i.e., the left-hand  
 492 side plate) uses the cubic spine function [16] and a 3×11 nodal arrangement. In this

493 arrangement, the assembly of discs falls into the choices of the disc–segment and disc–  
494 point contacts, depending on the disc locations as discussed in Figure 5.

495         The normal contact force at point *C* (i.e., the contact of discs 4 and 5) is  
496 examined. The results of the contact force are presented in Figure 14. The results  
497 include the simulations provided by the EFG–DE method, DEM, and Cundall and  
498 Strack [36]. In the DEM simulation, the plates were modeled as rigid walls and the test  
499 was reproduced using *PFC* programming. Excellent agreement is attained on both tests.  
500 In Test 1 however discrepancies occur in the early- to middle stage (i.e., before the 80<sup>th</sup>  
501 cycle). This means that the discs contact force is sensitive to the loading rate and the  
502 plate modulus. Specifically, when the rate is as low as in Test 2, the discrepancies fade  
503 off at the 20<sup>th</sup> cycle and are relatively small compared to the results arising from the  
504 rigid plate based simulations (i.e. the DEM and Cundall and Strack [36]). Where the  
505 rate is tripled as in Test 1, the discrepancies increase in amplitude and extension,  
506 meaning a stronger dynamic response. It is plausible to suggest that stronger dynamic  
507 responses of contact force occur where the loading rate is further increased or the plate  
508 modulus is decreased. Although the discrepancies exist, the trendline agreement on the  
509 two tests verifies the capability of the coupling method in approximating the dynamic  
510 response occurred between the continuum and granular domains.



511

512 Figure 14. Normal force at contact *C* in standard unit versus simulation cycle

513 determined by different simulation methods.

514

515 FE-DE simulations provide some new outcomes. It appears that Newmark  $\beta$

516 method was only conditionally stable in the FE domain, probably due to the relatively

517 large mesh sizes used. The allowable maximum time step  $\Delta t$  was coupling method

518 dependent. For example, given the modulus, the time step for a stable simulation is  $\Delta t$

519 = 0.01525 for EFG-DE coupling and is reduced to  $\Delta t = 0.00001525$  for FE-DE

520 coupling. It appears that EFG-DE program was stable under a larger time step. Even so,

521 the FE method offers advantage in simulation efficiency in simulating small

522 displacement problems, partially due to it updating interfacial contact forces by means

523 of movable loads [47]. This offers simplicity as opposed to the interaction adopted in

524 the EFG-DE coupling. On large-scale displacement problems, FE simulation cost can

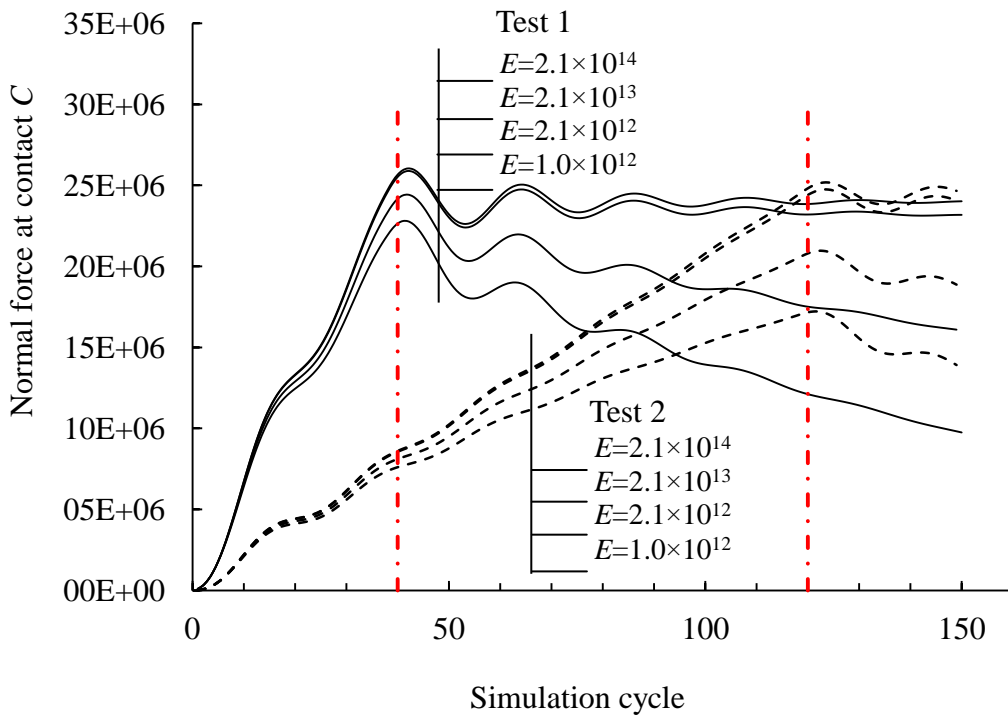
525 escalate due to re-meshing requirements as per Liu and Gu (2005).

526

527 *4.2.3. Influence of Young's modulus*

528 To gain an insight into the effects of plate deformation on the discs contact force,  
529 additional EFG–DE simulations were performed on scenarios where Young's modulus  
530 for the left-hand side plate was varied. Where the modulus is small, a large contact  
531 overlap tends to occur, and the results likely become unstable which is called *contact*  
532 *buckling* [37]. In this circumstance, as pointed out by Kanto and Yagawa [48],  
533 numerical oscillation may occur at contact because of the discontinuous velocity and  
534 acceleration when enforcing geometric compatibility. To prevent a severe contact  
535 overlap, Young's modulus was trialed and assigned  $E = 2.1 \times 10^{14}$ ,  $2.1 \times 10^{13}$ ,  $2.1 \times 10^{12}$ ,  
536 and  $1 \times 10^{10}$  units respectively for the plate. Similarly, Tests 1 and 2 that were used in  
537 the *nine disc test* were performed to examine the effects of the simulation cycles on the  
538 results. The results of the normal force at contact  $C$  obtained in the two tests under the  
539 varying plate modulus conditions are presented in Figure 15. In either test, the plate  
540 modulus noticeably influences the development of the contact force. The higher the  
541 modulus is, the greater the contact force will be. This relationship is more pronounced  
542 in stage two of the tests, i.e., the period when the plates stop moving and compressing  
543 the assembly of discs. Where the plate stiffness is relatively high, i.e.,  $E \geq 2.1 \times 10^{13}$   
544 units, the result curves coincide and approach equilibrium at the end of simulations.  
545 This trendline agrees with the results obtained in the DEM simulation (Figure 14). This  
546 means that the plate modulus of  $E \geq 2.1 \times 10^{13}$  units is high enough to satisfy the rigid

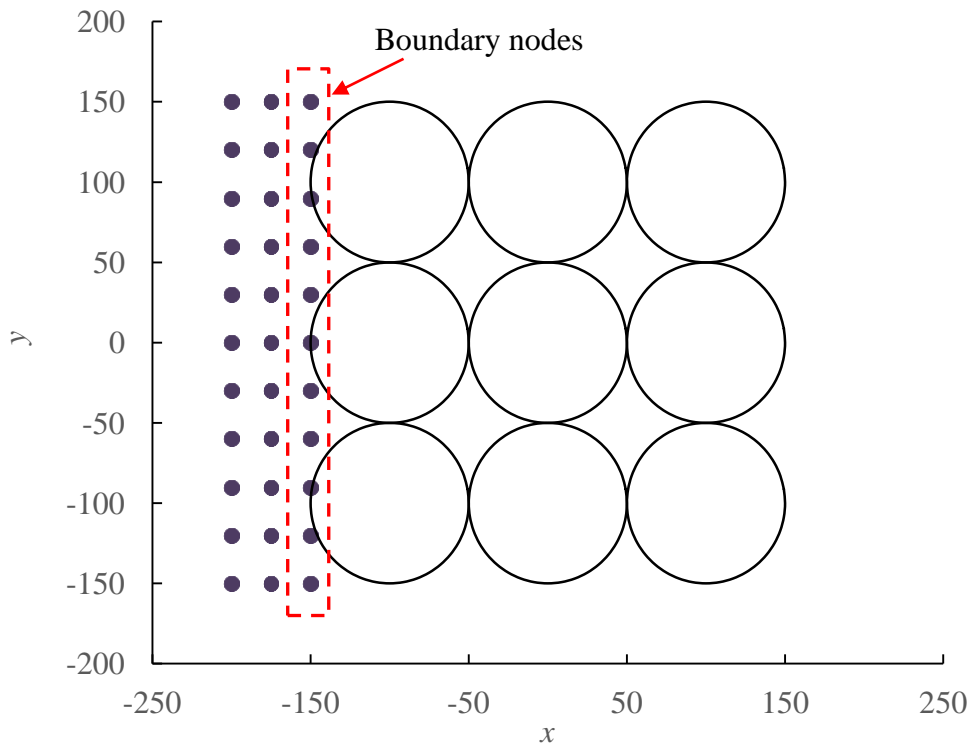
547 assumption made in the DEM simulation. Where the plate is less stiff, the contact force  
 548 attenuates over time. This is probably caused by the discs penetrate the plate when the  
 549 plate deforms, decreasing the overlap at contact C.



550  
 551 Figure 15. The relationship between the contact force in standard unit and the cycles  
 552 for test scenarios that assign the plates with varying Young's moduli in standard unit.

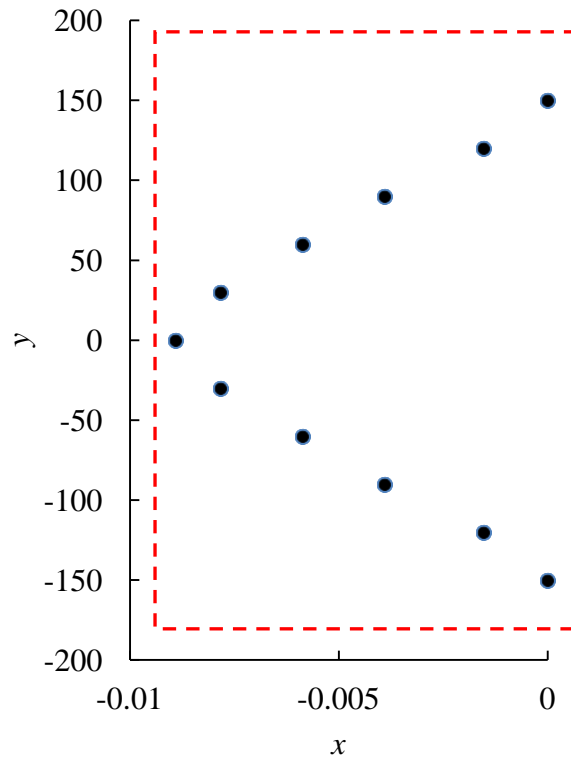
553  
 554 To gain a further insight into the response of a less stiff plate (i.e.,  $E = 1.0 \times 10^{12}$   
 555 units), the deformation occurred to the boundary nodes of the plate as shown in Figure  
 556 16 is examined. An enlarged view of the nodes displacement captured at the 40<sup>th</sup> cycle  
 557 is shown in Figure 17. Due to the use of the deformable plate, the actual displacement  
 558 at the plate boundary is not uniform. The central nodes displace significantly greater

559 than those occurred on the upper and bottom plates. The location dependency agrees  
560 with the observations occurred in the tri-axial tests [42, 49] where the central section of  
561 the samples dilated and thus presented greater deformation.



562

563 Figure 16. Boundary nodes location on the plate–discs interface examined for the  
564 plate deformation.



565

566 Figure 17. Displacement in standard unit of the boundary nodes on the plate-discs

567 interface recorded at the 40<sup>th</sup> cycle.

568

## 569 5. CONCLUSIONS

570 This paper presents an element-free, multiscale EFG-DE coupling method. This

571 method is developed to simulate multibody interactions, in particular, the continuum-

572 granular contact problems. This method uses a transient disc-segment contact

573 algorithm for the contact problems and is applied to two example problems. This study

574 arrives at the following conclusions.

575 The coupled EFG-DE method is free of meshing or re-meshing, thus enabling

576 reasonable computation costs and stable calculation. This method applies the



577 Newmark- $\beta$  method to the continuum and the central difference method to the granular  
578 domain to solve the dynamic problem in a discrete form. This method uses an explicit–  
579 implicit time scheme and attains satisfactory computation stability. This method  
580 develops a transient contact detection algorithm which enables accurate, seamless force  
581 exchange on the domains interface, and accounts for deformable boundaries. The  
582 method is applied to two example problems and verified against the existing analytical  
583 and simulation results, thus confirming its capabilities in simulating dynamic  
584 interaction occurred between continuum and granular media. As opposed to other  
585 coupled numerical approaches (e.g., FEM–DEM or FEM–EFGM), this current method  
586 is capable to conduct numerical analysis with less external interventions. In addition  
587 the method is able to separate different domains and examine the particular behaviour  
588 of interest. It is envisaged that the proposed method will be applied to large-scale, multi-  
589 body interaction problems to further verify its performance.

590

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596

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